



# SERVICE LETTER

ID: **AEF-0964-SLE-0004**

Issue: **01**

Date: **07-07-2025**

Subject: **HEC\_5\_H125**  
**New end fitting configurations**

EASA STC no.: **10085893**

STC parts S/N **All**

ATA chapter **n/a**

DOA reference: **EASA.21J.644**

Aircraft manufacturer: **AIRBUS Helicopters**

Aircraft type: **AS350 B3 (H125)**

S/N: **All**

**To all Customers, Operators and Service Centers**

This Service Letter is issued to draw attention to the following information:

The flight manual supplement AEF-0819-FMS-0005, Ref [1] for the HEC\_5\_H125 complex PCDS (personnel carrying device systems) and the instructions for continued airworthiness AEF-0819-ICA-0007, Ref [2] have been amended to Issue -03 to incorporate new end fitting variants in place of the rescue hook HUB 7/8-8 (standard end fitting). The new end fitting variants are the carabiner KH-301-SSB1 and the oblong ring AW-13-10. All end fitting variants can be used in a configuration where they are engaged directly to the load element, without sharp edge protection steel rope in between. The new end fitting variants are optional. There is no obligation to retrofit the existing HEC\_5\_H125 PCDS. The new end fitting variants can be ordered at Airworks & Heliseilerei GmbH. The flight manual supplement and the instructions for continued airworthiness do not change in terms of limitations.

The instructions for continued airworthiness AEF-0819-ICA-0007, Ref [2] have been amended to issue -03 to give supplemental instructions for the use of the rescue hook HUB 7/8-8 and to give guidance for compatibility of the rescue hook HUB 7/8-8 and other end fitting variants with other personnel carrying device systems (PCDS). The instructions for continued airworthiness do not change in terms of limitations.

Note that the operator is responsible for selection of conform PCDS that is engaged into the rescue hook HUB 7/8-8 (standard end fitting) or into other approved variants of the end fitting.

For further information on this subject, please contact the following address:

**DO EASA 21J.644**

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**Attachments:**

Ref [1]	Flight Manual Supplement, HEC_5_H125	AEF-0819-FMS-0005-03
Ref [2]	Instructions for continued airworthiness, HEC_5_H125	AEF-0819-ICA-0007-03