

## SUPPLEMENTAL TYPE CERTIFICATE

**10085893**

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### AIRWORK & HELISEILEREI GmbH

BAHNHOFWEG 1  
6405 IMMENSEE  
SWITZERLAND

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** EASA.R.008

**Type Certificate Holder:** AIRBUS HELICOPTERS

**Type:** AS 350 / EC 130

**Model:** AS 350 B3

**Description of Design Change:**

Complex PCDS for dual hook HEC system.

**EASA Certification Basis:**

The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:

the following paragraph(s) at a later amendment: CS 27.865 Amdt 10

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

**Associated Technical Documentation:**

Rotorcraft Flight Manual Supplement, AEF-0819-FMS-005-01, dated 18 Nov 2024

Instructions for continued airworthiness, AEF-0819-ICA-007, dated 04 Nov 2024

See Continuation Sheet(s)

**For the European Union Aviation Safety Agency**

**Cologne, Germany, 26 November 2024**

**Fabrice LEGAY**  
**Section Manager**  
**Medium & Light Rotorcraft**





Certification Program, AEF-0819-PCP-0023-05, dated 20 Nov 2024  
or later revisions of the above listed document(s) approved/accepted under the EASA system.

**Limitations/Conditions:**

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

